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Distribution of the chord of blade profiles and

wings

The distribution of the profiles chords defining the geometry of our Propeller or wind turbine's blades Is an important determinant of performance of our wind turbine blade, Hydroelectric, boat propeller, propeller aircraft or fan ... In the dimensions tab, a series of sliders allows you to model the shape of your blade or wing in all its dimensions.



We have seen with <u>wings</u> that leaks wingtip generate a loss of <u>lift</u> and induced drag that is detrimental to the <u>performance</u> or <u>lift/drag ratio</u> of our wing.



Our <u>blades of wind turbine</u> or <u>propeller</u> are wings (in rotation), it is natural that this phenomenon of losses at blade tip is also taken into account in the <u>form of our blade</u>. We have seen in the <u>description of the aspect ratio</u> that the losses are greater if the wing aspect ratio low.

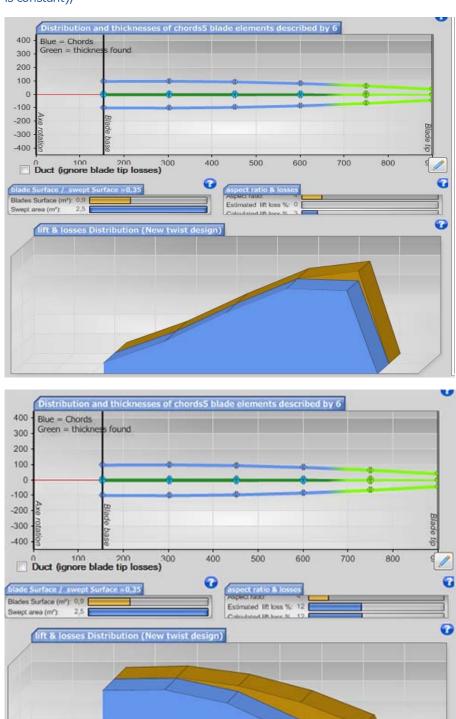
The distribution cords must respect a sufficient <u>aspect ratio</u> to avoid wasting energy. An helicopter blade has a significant aspect ratio, blade's tip losses are thus very small. The propeller or wind turbine's blades, will be the longest possible, to reduce losses, but in practice we are often limited in length, so the distribution of the lift must concentrated as far as the blade's tip. The shape making the most of this condition for the wing is the ellipse, but t we have already discussed this in the <u>overview article about wings</u>

For a <u>rotary wing</u>, as the <u>propeller blade</u>, the speed received by the <u>elements of the blade</u> is composed of the rotational speed, and the speed of the fluid to the impeller. This implies that the perceived speed of an element at the blade tip is higher than the rate seen by an element dose to the propeller shaft (<u>apparent velocity</u>).

The formula for the lift: **Lift**(2D)(in Newtons)= Cz X wing area(m^2) X fluid density (Kg/m3) X (**speed**²(m/sec)/2).

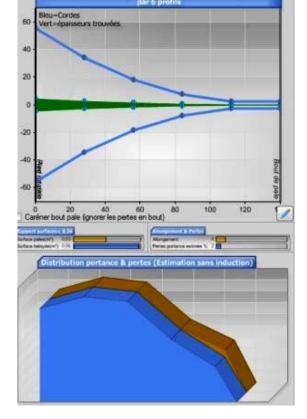
The lift is directly related to the square of the speed seen by the profile . if we do not take into account losses at the <u>wing</u> tips, the lift changes with the square of the speed. Therefore equal distribution of <u>chords</u> on a rotating wing will focus the lift force on the fastest elements, so towards the blade's tip.

the same shape but for a <u>wing</u>. This clearly shows that the distribution of lift is centered on the end to the propeller (because the relative velocity is higher in the blade tip) whereas the wing lift distribution follows the distribution of <u>chords</u> (as the relative speed is constant);



Distribution of <u>chords</u> and lift, 5 <u>blade elements</u> and wing (blue with losses at the <u>blade</u> tip) of a propeller (picture above) and a <u>wing</u> (bottom image) form developed identical We have seen that the best distribution of the lift was elliptical.

For <u>wings</u>, an elliptical distribution of <u>chords</u> gives elliptical distribution of the lift. But a <u>propeller</u> if it is desired that the lift is distributed elliptically we need to increase the strings of elements dose to the propeller shaft to compensate the difference of relative speeds at the elements:



Chords and elliptic lift distribution of a propeller

Comparing a commercial <u>Wind turbine blades</u> whose shapes are optimized to focus the lift farthest from the tip in order to reduce losses of the blade tip, and a wing shape optimized for the same effect:





So be sure to spread your chords distribution to reduce losses to a minimum blade tip and increase the <u>performance</u> of your <u>propellers</u>. For a pusher propeller the concept of performance requires clarification, which takes into account the <u>propulsive efficiency</u>

See also the propeller blades and wind turbines

This is a determinant $\underline{\text{factor of cavitation for propellers and hydrofoils}}$ in liquid environment .

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